

Fig. 3. Experimental setup. Up: front view, down: top view.

During the tests, the dry bulb temperature inside the tunnel varied between 22.3 °C and 26.6 °C; the tunnel wet bulb temperature between 16 °C and 22 °C; and the laboratory barometric pressure between 1012.81 hPa and 1023.84 hPa. These changes in the variables correspond to the natural environmental changes recorded during five days of testing. Once the barometric pressure, dry and wet bulb temperature are measured for each operating condition, the relative humidity  $\phi$  can be determined by means of the Ferrel equation [13]:

$$\phi = \frac{P_{vs}(T_v) - \phi \cdot P_B \cdot (T - T_v)}{P_{vs}(T)} = \frac{P_{vs}(T_v) - \phi_0 \cdot (1 + 0.00115 \cdot (T_v - 273.15)) \cdot P_B \cdot (T - T_v)}{P_{vs}(T)}, \quad (2)$$

where  $T_v$  is the absolute wet bulb temperature (K),  $\phi$  is the so-called psychrometric constant pertinent to the standard wet-bulb temperature,  $\phi_0$  is the so-called psychrometric constant pertinent to the standard wet-bulb temperature of 0 °C, which was experimentally adjusted by Ferrel. Table 1 provides the values of the coefficients in Eqs. (2)-(4), (6)-(7),  $P_B$  is the barometric pressure, and  $T$  is the absolute temperature (K). Eq. (3) is used to obtain the saturated vapor pressure,  $P_{vs}$ :

$$P_{vs}(T) = 1Pa \cdot e^{(\hat{A} \cdot T^2 + \hat{B} \cdot T + \hat{C} + \hat{D}/T)}. \quad (3)$$

Once the relative humidity is known, the moist air density for each operating condition can be obtained via the CIPM-2007 revised formula [14], which is valid for the temperature and barometric pressure ranges of this work:

$$\rho = \frac{P_B \cdot M_a}{Z \cdot R \cdot T} \cdot \left[ 1 - x_v \left( 1 - \frac{M_v}{M_a} \right) \right], \quad (4)$$

$R$  is the ideal gas constant,  $M_a$  is the molar mass of dry air and  $M_v$  is the molar mass of water. The following equations are used to determine: the mole fraction of water vapour  $x_v$ , the enhancement factor  $f$  and the compressibility factor  $Z$ :

$$x_v = \phi \cdot f(P_B, t) \cdot \frac{P_v(T)}{P_B}, \quad (5)$$

$$f = \alpha + \beta \cdot P_B + \gamma \cdot t^2, \quad (6)$$

$$Z = 1 - \frac{P_B}{T} \cdot [a_0 + a_1 \cdot t + a_2 \cdot t^2 + (b_0 + b_1 \cdot t) \cdot x_v + (c_0 + c_1 \cdot t) \cdot x_v^2] + \frac{P_B^2}{T^2} \cdot (d + e \cdot x_v^2), \quad (7)$$

where  $t$  is the air-dry bulb temperature expressed in °C and  $P_v$  is the partial vapor pressure (Pa).

In Section 3, a dimensional analysis based on the Reynolds number is performed. Thus, the determination of the moist air dynamic viscosity  $\mu$  is required. This property can be calculated via the theoretical formulation of Mason & Monchick [15], which was experimentally validated by Kestin & Whitelaw [16] by means of an oscillating disc viscometer, and it is recommended by “NASA Langley Research Centre” [17] for outdoor-indoor air applications considering atmospheric pressure, a temperature between 10 °C and 50 °C, and a relative humidity in the range from 0.3% to 92%:

$$\mu = \alpha_0 + \alpha_1 \cdot T + (\alpha_2 + \alpha_3 \cdot T) \cdot x_v + \alpha_4 \cdot T^2 + \alpha_5 \cdot x_v^2, \quad (8)$$

According to the obtained results, the relative humidity during the tests was always within the interval 37.4%-66.8%; the moist air density within the interval 1.1738 kg/m<sup>3</sup> - 1.1982 kg/m<sup>3</sup>, which means an air density variation equal to 2.44 g/m<sup>3</sup>; and the dynamic viscosity remained almost constant since the measures were within the interval from 1.82 · 10<sup>-5</sup> Pa · s to 1.84 · 10<sup>-5</sup> Pa · s.

The anemometer rotation frequency, for each tested wind speed, is recorded with a sampling frequency of 1 Hz during 50 s, so that steady-state conditions can be considered. The cup anemometer rotation frequency  $f_r$  is calculated as the mean value of the  $m$  measurements recorded:

$$f_r = \frac{1}{m} \sum_{j=1}^m f_{rj}. \quad (11)$$

**Table 1.-** Constant coefficients for the determination of environmental properties.

Constant terms		Value	Units	Notes	
Ferrel Eq. (2)	$\alpha_0$	$6.6 \cdot 10^{-4}$	$^{\circ}\text{C}^{-1}$	Adjusted by Ferrel	
Vapour pressure Eq. (3)	$\hat{A}$	$1.2378847 \cdot 10^{-5}$	$\text{K}^{-2}$	Specified by P. Giacomo [18]	
	$\hat{B}$	$-1.9121316 \cdot 10^{-2}$	$\text{K}^{-1}$		
	$\hat{C}$	33.93711047			
	$\hat{D}$	$-6.3431645 \cdot 10^3$	$\text{K}$		
Moist air density Eq. (4)	<i>Ideal gas constant</i>	$R$	$8.314472$	$\text{J} \cdot \text{mol}^{-1} \cdot \text{K}^{-1}$	Recommended by Codata 2006 [19]
	<i>Molar mass of dry air</i>	$M_a$	$28.96546 \cdot 10^{-3}$	$\text{kg} \cdot \text{mol}^{-1}$	It is assumed a background of 400 $\mu\text{mol} \cdot \text{mol}^{-1}$ for the mole fraction of carbon dioxide in air
	<i>Molar mass of water</i>	$M_v$	$18.01528 \cdot 10^{-3}$	$\text{kg} \cdot \text{mol}^{-1}$	
Enhancement factor Eq. (6)	$\alpha$	1.00062		Specified by P. Giacomo [18]	
	$\beta$	$3.14 \cdot 10^{-8}$	$\text{Pa}^{-1}$		
	$\gamma$	$5.6 \cdot 10^{-7}$	$\text{K}^{-2}$		
The compressibility factor Eq. (7)	$a_0$	$1.58123 \cdot 10^{-6}$	$\text{K} \cdot \text{Pa}^{-1}$	Specified by P. Giacomo [18]	
	$a_1$	$-2.9331 \cdot 10^{-8}$	$\text{Pa}^{-1}$		
	$a_2$	$1.1043 \cdot 10^{-10}$	$\text{K}^{-1} \cdot \text{Pa}^{-1}$		
	$b_0$	$5.707 \cdot 10^{-6}$	$\text{K} \cdot \text{Pa}^{-1}$		
	$b_1$	$-2.051 \cdot 10^{-8}$	$\text{Pa}^{-1}$		
	$c_0$	$1.9898 \cdot 10^{-4}$	$\text{K} \cdot \text{Pa}^{-1}$		
	$c_1$	$-2.376 \cdot 10^{-6}$	$\text{Pa}^{-1}$		
	$d$	$1.83 \cdot 10^{-11}$	$\text{K}^2 \cdot \text{Pa}^{-2}$		
	$e$	$-0.765 \cdot 10^{-8}$	$\text{K}^2 \cdot \text{Pa}^{-2}$		
The dynamic viscosity Eq. (8)	$\alpha_0$	$8.4986 \cdot 10^{-7}$	$\text{Pa} \cdot \text{s}$	Determined by Mason & Monchick. Recommended by NASA [17]	
	$\alpha_1$	$7 \cdot 10^{-8}$	$\text{Pa} \cdot \text{s} \cdot \text{K}^{-1}$		
	$\alpha_2$	$1.13157 \cdot 10^{-6}$	$\text{Pa} \cdot \text{s}$		
	$\alpha_3$	$-1 \cdot 10^{-8}$	$\text{Pa} \cdot \text{s} \cdot \text{K}^{-1}$		
	$\alpha_4$	$-3.7501 \cdot 10^{-11}$	$\text{Pa} \cdot \text{s} \cdot \text{K}^{-2}$		
	$\alpha_5$	$-1.00015 \cdot 10^{-6}$	$\text{Pa} \cdot \text{s}$		

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In the middle of the rotation frequency sampling interval, the dynamic pressure is recorded by means of the Pitot tube, which provides the mean value of  $l = 20$  measurements obtained during 10 s previous to trigger it. The mean wind dynamic pressure  $\Delta p$  at the anemometer position for each operating condition is corrected through several factors detailed below:

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$$\Delta p = \frac{k_c}{C_h} \cdot k_f^2 \cdot \frac{1}{l} \sum_{k=1}^l \Delta p_i. \quad (9)$$

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The comparison between measurements from the Pitot tube placed at the reference position and at the anemometer position (Fig. 3) allows obtaining the wind tunnel calibration factor  $k_c$  and its uncertainty. The alignment accuracy of the Pitot tube with the wind flow direction is determined via the Pitot tube head coefficient  $C_h$ . In the present study, the value of this coefficient is determined according to the recommendations given by the ISO 3966, which deals with measurement of fluid flow velocity using Pitot static tubes [20]. Finally, the influence of the anemometer shape on the mean flow field velocity  $V_b$  is quantified by the blockage correction factor  $k_f$  calculated by applying the Maskell theorem [21]:

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$$k_f = \frac{V_b}{V} = 1 + \frac{1}{2} a \cdot c_f \cdot b, \quad (12)$$

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where  $b$ , the blockage ratio, is the ratio of front area to the tunnel cross section; in our case  $b = 0.025$ . According to the recommendations of Barlow et al. [22], regarding unusual shapes tested in a wind tunnel, the product of the shape factor  $a$  and the force coefficient  $c_f$  is 0.5. The value of  $k_f$  is experimentally checked in our wind tunnel by comparing the measurements provided by the Pitot tube placed at the anemometer section with and without the cup anemometer inside the tunnel.

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Finally, Eq. (10), which is obtained from Bernoulli's equation, provides the wind speed  $V$  at the anemometer location:

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$$V = \sqrt{\frac{2 \cdot \Delta p}{\rho}}, \quad (10)$$

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The uncertainties of the different magnitudes are obtained following the Guide to the Expression of Uncertainty, commonly referred to as GUM [23]; besides that, the methodology employed by CENAM [24] is useful to determine the uncertainty of the moist air density and viscosity.

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### 3. Results and discussion

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#### 3.1. Direct measurements and dimensionless abacus

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Fig. 4 shows the  $f_r$  and  $V$  measures accomplished in the wind tunnel. It is remarkable the variations observed in  $f_r$  for a specific  $V$ , this means a low repeatability due to ambient conditions variability. Wind speed increases those variations from indistinguishable differences at low  $V=1.6$  m/s to a 15 rad/s difference for  $V=17$  m/s.

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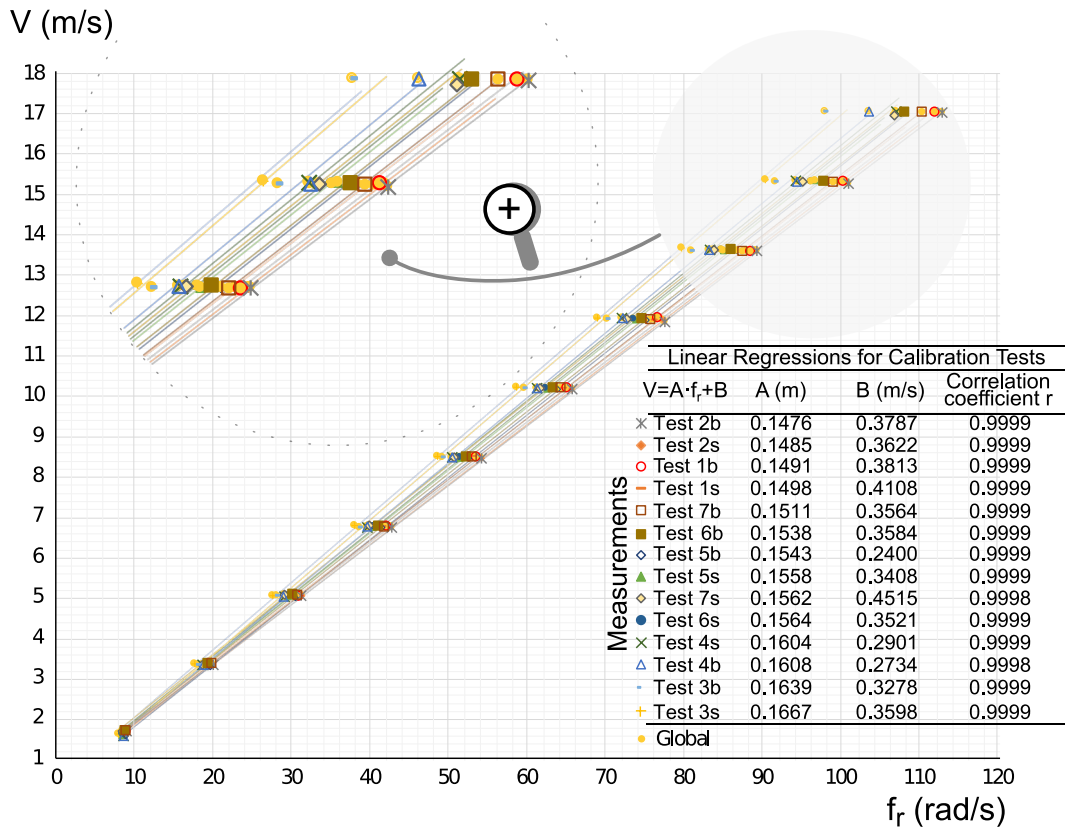


Fig. 4. Fourteen calibrations performed on the same cup anemometer at different ambient conditions.

In order to explain the discrepancies observed in the slope of the calibration lines in Fig. 4, the mean moist air density is calculated for each experimental test (Tests 1s-7s and Tests 1b-7b). The slope coefficients  $A$  are represented in Fig. 5 respect to their corresponding densities  $\rho$ . The large dispersion observed in each point ( $\rho$ ,  $A$ ), considering 95% confidence level, demonstrates that is difficult to fit  $\rho$  and  $A$  values by a linear regression, hereinafter referred to as “direct method”.

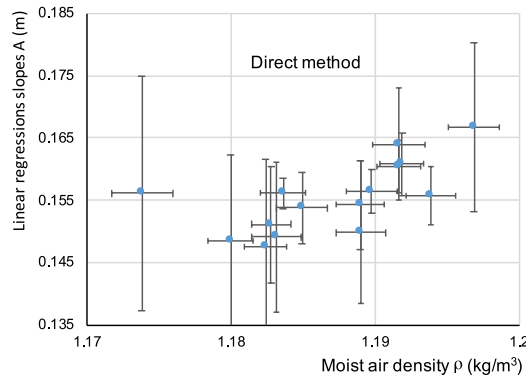


Fig. 5. Moist air density versus anemometer calibration slope: Direct Method. Bars represent a 95% confidence level.

In order to overcome the large uncertainties showed in Fig. 5, a dimensional analysis is conducted (Fig. 6). Applying the Buckingham Pi Theorem, the  $f_r - V$  domain is mapped into a new domain  $\Pi_1 - \Pi_2$ :

$$\Pi_1 = Re(f_r) = \frac{\rho \cdot f_r \cdot D^2}{\mu}, \quad (11)$$

$$\Pi_2 = TSR^{-1} = \frac{V}{f_r \cdot D}, \quad (12)$$

Reynolds number  $Re$  measures the ratio of inertia forces to viscous forces in a flow [25], and the Tip Speed Ratio  $TSR$  is the relation between the cup velocity and the free stream air speed.

The resulting “dimensionless abacus” is presented in Fig. 6. In this domain, the regression lines correspond to each tested wind speed, while the set of hyperbolic curves describes the measurements evolution respect to the inverse of the kinematic viscosity. Regression and correlation coefficients, as well as the correspondence of the hyperbolic lines with the moist air density (considering the dynamic viscosity quasi-constant during the experimental tests) are included in Fig. 6. Note that, the relative uncertainties of  $Re(f_r)$  have low values. On the other hand, small density changes (in the order of hundredths of a  $kg/m^3$ ) can modify the cup anemometer rotation frequency.

217 Considering a linear relation  $V=A_i f_r + B$ , such as in the direct method, and the dimensionless abacus, it is possible to determine  $A$   
 218 value as a function of the inverse of the kinematic viscosity  $\nu^{-1}$  and  $f_r$ . Once the regression lines are determined, these lines and  
 219 the Eqs. (11)-(12) provide the rotation frequency  $f_{r,i}$  and  $\nu_i^{-1}$  for specific Reynolds  $Re_i$  and wind speed  $V_i$ :

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$$\frac{1}{TSR_i} = Re_i/a_i + b_i, \quad (13)$$

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$$f_{r,i} = \frac{TSR_i}{D} \cdot V_i, \quad (14)$$

222 
$$\nu_i^{-1} = \rho_i/\mu = \frac{Re_i}{f_{r,i} \cdot D^2}. \quad (15)$$

223 Finally, the slope coefficient  $A_i$  corresponding to  $f_{r,i}$ , and therefore to  $\nu_i^{-1}$  is:

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$$A_i = \frac{V_i - B}{f_{r,i}}. \quad (16)$$

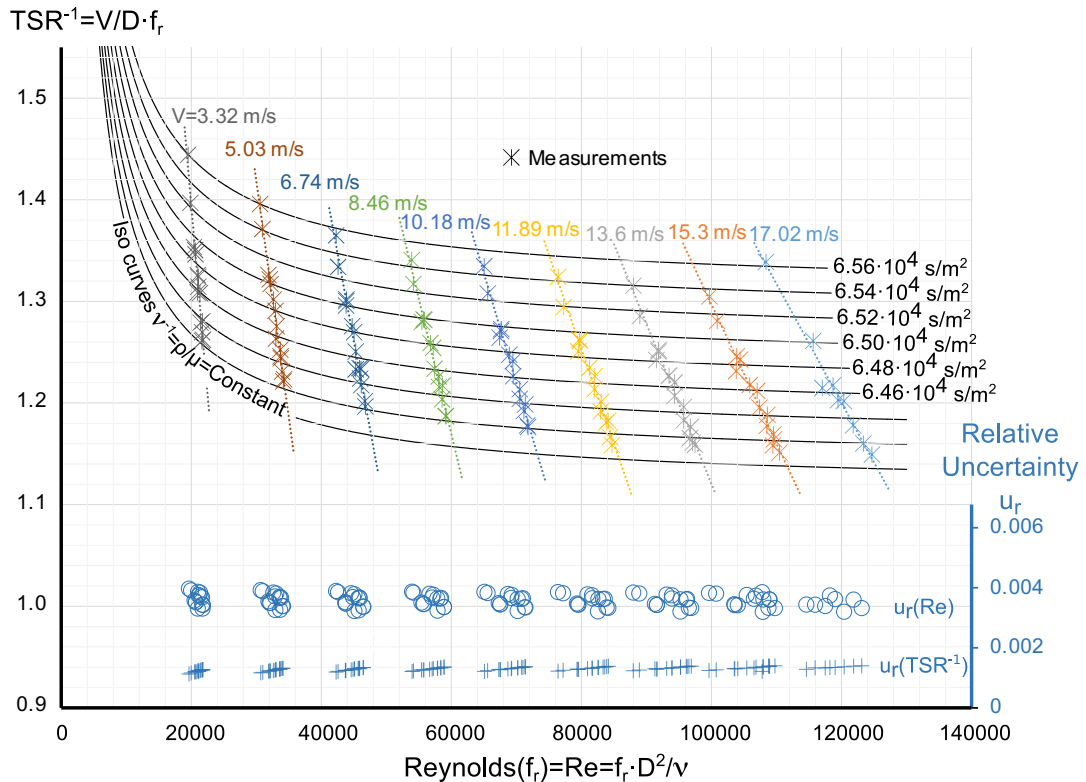
225 The following assumptions have been considered: firstly, if air density remains nearly constant during a calibration test, kinematic  
 226 viscosity remains constant as well, and consequently, it is possible to determine a constant value for  $A$ ; and secondly, the intercept  
 227 coefficient  $B$  does not depend on kinematic viscosity (all calibration lines in Fig. 4 seem to have a similar intercept). The dynamic  
 228 viscosity  $\mu$  remains quasi-constant in the present research and in many practical applications, so it is easy to compute  $\rho_i$  from  
 229  $\nu_i^{-1}$ .

230 The dimensional analysis provides the  $\rho$ - $A$  values portrayed in Fig. 7 and the following linear model that estimates  $A$  for a specific  
 231  $\rho$ :

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$$A = c \cdot \rho - d. \quad (17)$$

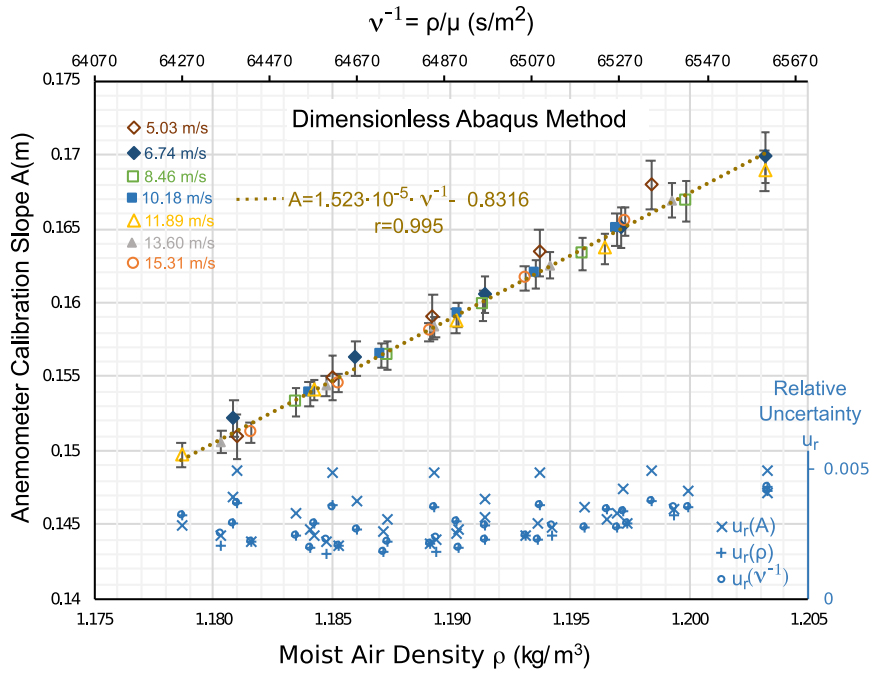
233 The observed low dispersions, at 95% confidence interval, explain the high determination coefficient  $R^2=0.995$  for the linear  
 234 regression (Eq. (17)) provided by the hereafter called Dimensionless Abacus Method, compared to other studies [1], which use  
 235 a Direct Method. Fig. 7 also shows the relative uncertainties of the calibration slope, moist air density and kinematic viscosity.

236 Eq. (17) shows how calibration lines change with the variation of air density at the laboratory, so that if air density increases, the  
 237 slope of the calibration line increases as well, and vice versa.  
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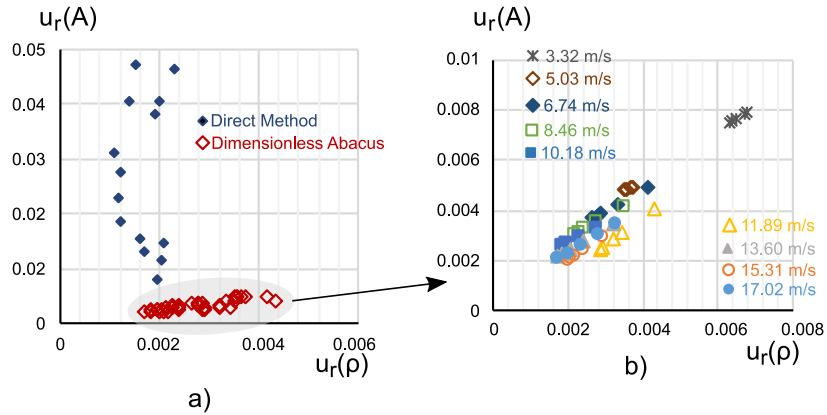
239 **Fig. 6.** Dimensionless abacus  $Re(f_r) - TSR^{-1}$ . Experimental values for several wind speeds at different ambient conditions  
 240 (different air flow viscous forces).  
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244 **Fig. 7.** Moist air density versus anemometer calibration line slope: Dimensionless abacus method. Bars represent a 95%  
245 confidence level.

246 Relative uncertainties are compared in Fig. 8.a. It is noteworthy the low and almost constant values provided by Dimensionless  
247 Abacus Method compared with Direct Method. Fig. 8.b classifies the relative uncertainties using the Dimensionless Abacus  
248 Method respect to wind speed. A reduction in relative uncertainty  $u(A)/A$  is observed as the wind speed increases, this is a  
249 consequence of the manometer uncertainty, which grows as wind velocity decreases.



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251 **Fig. 8.** Relative uncertainty of anemometer calibration line slope: a) Dimensionless Abacus Method versus Direct Method, b)  
252 Dimensionless Abacus Method values at tested wind speeds.

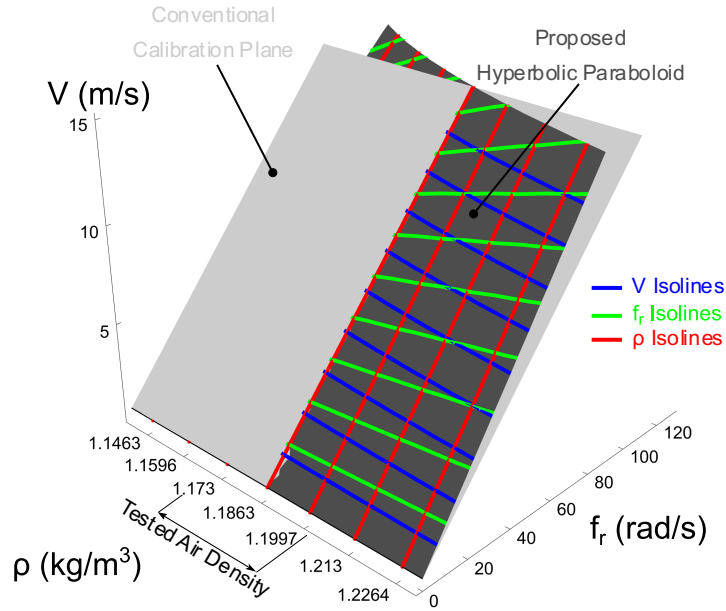
253 Note that, with the exception of  $V=3.32$  m/s, the values represented in Fig. 7-8 lay within the wind speed range recommended by  
254 the annex F of the IEC international standard [11]. This range is established between 4 m/s and 16 m/s. According to the tests  
255 accomplished, the uncertainty is high for  $V < 4$  m/s, and the linear Eq. (17) changes for  $V > 16$  m/s.

### 256 257 3.2. The hyperbolic paraboloid surface.

258 The Eq. (17), obtained using the Dimensionless Abacus Method, helps to estimate  $A$  respect to  $\rho$  and therefore, the anemometer  
259 calibration Eq. (1) can be rewritten for  $V \in [4 \text{ m/s}, 15.3 \text{ m/s}]$  as follows:

$$260 \quad V = (c \cdot \rho - d) \cdot f_r + B, \quad (18)$$

261 Eq. (18) is a hyperbolic-paraboloid surface: a doubly ruled surface, since it satisfies the conditions described for those quadratic  
262 surfaces [26]. According to Fig. 9, air density and rotation frequency isolines are the hyperbolic paraboloid surface rulings. In  
263 order to see the differences between the hyperbolic-paraboloid surface and the planar surface obtained by applying the IEC  
264 anemometry method [11], both surfaces have been extrapolated outside the measured density range (it is considered a mean  
265 density value during the calibration process equal to  $1.1863 \text{ kg/m}^3$ ). When the anemometer is used in field application where the  
266 air density differs from that at the calibration laboratory, the measurement provided by the anemometer varies. The IEC method  
267 does not take into consideration this fact.



**Fig. 9.** Moist air density effect on an Auriol IAN cup anemometer. Calibration surfaces: Plane from the IEC anemometry method [11] and Hyperbolic Paraboloid from the Dimensionless Abacus Method.

Above  $V = 15$  m/s the system does not behave as the ruled surface described by Eq. (18). A nonlinear response appears, which depends on the moist air density. It is noteworthy that, for air density measurements upper than  $1.1863 \text{ kg/m}^3$ , rotation frequency growths more slowly than if the air density is lower than  $1.1863 \text{ kg/m}^3$ .

Fig. 10 shows the  $\rho$  and  $f_r$  ruling lines as well as  $V$  isocurves (parabolas) in the  $V(\rho, f_r)$  projected views. Note how the surface rulings evolve to curvilinear paths above 15 m/s (Fig. 10 a and b). Moreover, the curvature is concave for high moist air density values, and convex for low moist air density values. This effect of air density (or kinematic viscosity) could explain the underspeed or overspeed of cup anemometers observed in some field measurements, such as those highlighted by Kristensen [27,28], although this author relates this behaviour to the effect of air turbulence.

According to Fig. 10.b, the slope of the rotation frequency iso-curves, up to  $V=15$  m/s, is a measure of how the wind speed changes  $\Delta V$  due to air density variation  $\Delta \rho$  during the IEC calibration process:

$$\frac{\Delta V}{\Delta \rho} = c \cdot f_r^*, \quad (19)$$

Eq. (19) depends on the anemometer and provides the slope of the rotation frequency iso-curves respect to: the cup anemometer rotation frequency considering the air density at the calibration laboratory  $f_r^*$ , and the “cup-anemometer air density sensibility” parameter  $c$  (Eq. (17)). For example, an Auriol IAN anemometer has a  $c = 0.8315 \text{ m}^4/\text{kg}$ , if  $f_r^* = 80 \text{ rad/s}$ , then a density variation of  $\Delta \rho = 0.013 \text{ kg/m}^3$  with respect to air density during the calibration process leads to  $\Delta V = 0.86 \text{ m/s}$ , which will grow proportionally with the increments of air density.

Finally, Fig. 11 portrays the wind tunnel experimental data, the hyperboloid paraboloid surface and the IEC calibration line. The proposed Dimensionless Abacus model has a maximum absolute error of 0.4 m/s while the IEC calibration line has a maximum absolute error of 0.9 m/s.

In order to consider the influence of air density on wind speed measurements, the IEC linear regression [11] can be corrected by means of Eq. (19). On the other hand, the hyperboloid paraboloid surface considers the ambient conditions, thus wind speed estimation does not need additional corrections. In this regard, and according to Fig. 12, if  $\rho > \rho_c$ , where  $\rho_c$  is the air density at the calibration laboratory and  $\rho$  is the air density in a specific location, then the wind speed model obtained from the Direct Method provides values below the actual measurements, while if  $\rho < \rho_c$  the Direct Method regression curve is above the real measurements. The hyperbolic parabolic regression does not need correction in any case.

The abacus showed in Fig. 6 and Eqs. (13), (15) point out that air-cup friction produces the aforementioned deviations. Hence, instead of air density, the kinematic viscosity is the most important ambient parameter. It is adequate the use of the moist air kinematic viscosity for wide ranges of temperature and density, such as those considered in Annex I of the IEC standard [11]: from  $0.9$  to  $1.35 \text{ kg/m}^3$  (classifications A, B, C and D). For these cases, it is not possible to establish an equivalence between the air density and the inverse of the kinematic viscosity. Therefore, according to Fig. 7, Eqs. (17)-(19) should be rewritten respect to the inverse of the moist air kinematic viscosity as follows:

$$A = e \cdot \frac{1}{\nu} - d, \quad (20)$$

$$V = (e \cdot \frac{1}{\nu} - d) \cdot f_r + B, \quad (21)$$

$$\frac{\Delta V}{\Delta(1/\nu)} = e \cdot f_r^*, \quad (22)$$

the parameter  $e$  is the “cup-anemometer air kinematic viscosity sensibility”, and  $f_r^*$  is the equivalent rotation frequency when in-field air kinematic viscosity is equal to that of the calibration laboratory.

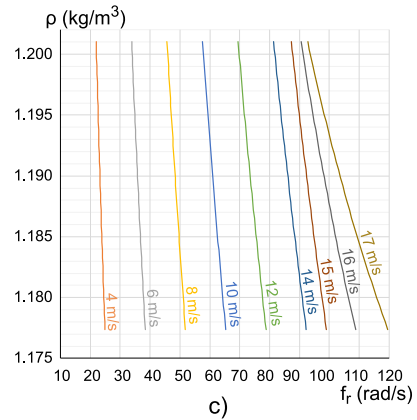
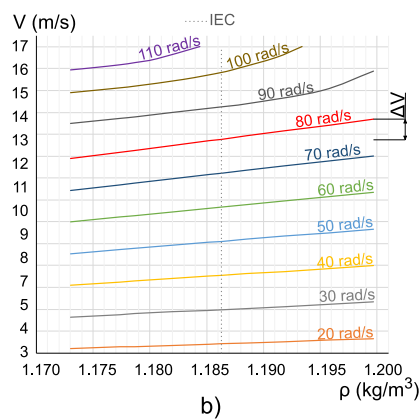
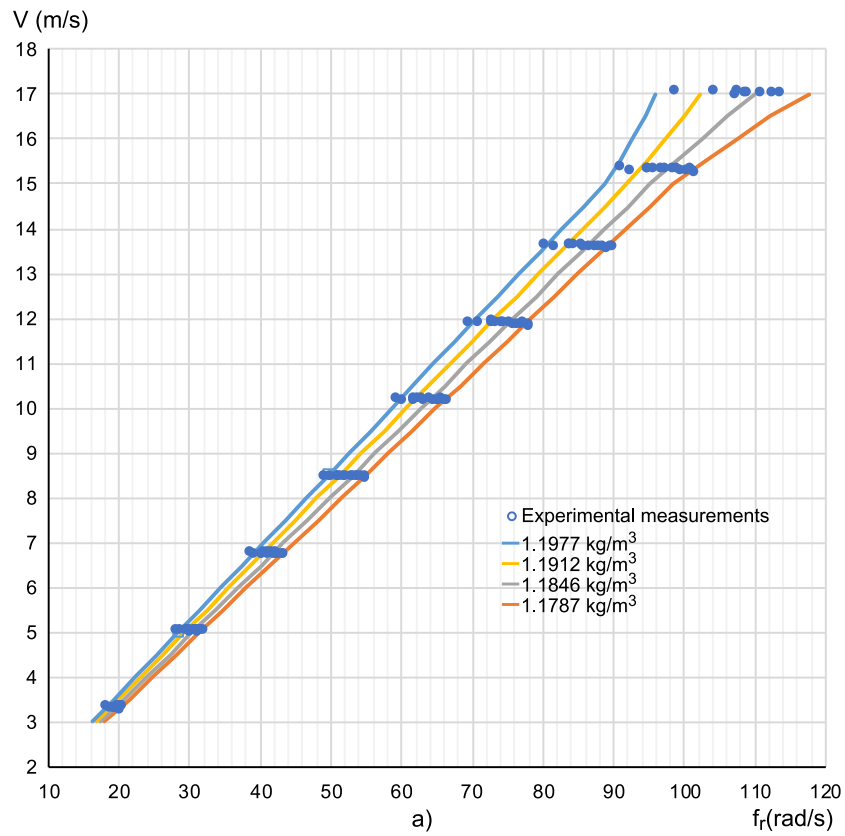


Fig. 10. Projected views of the Auriol IAN calibration surface. a) Front view, b) Top view, c) Side view.

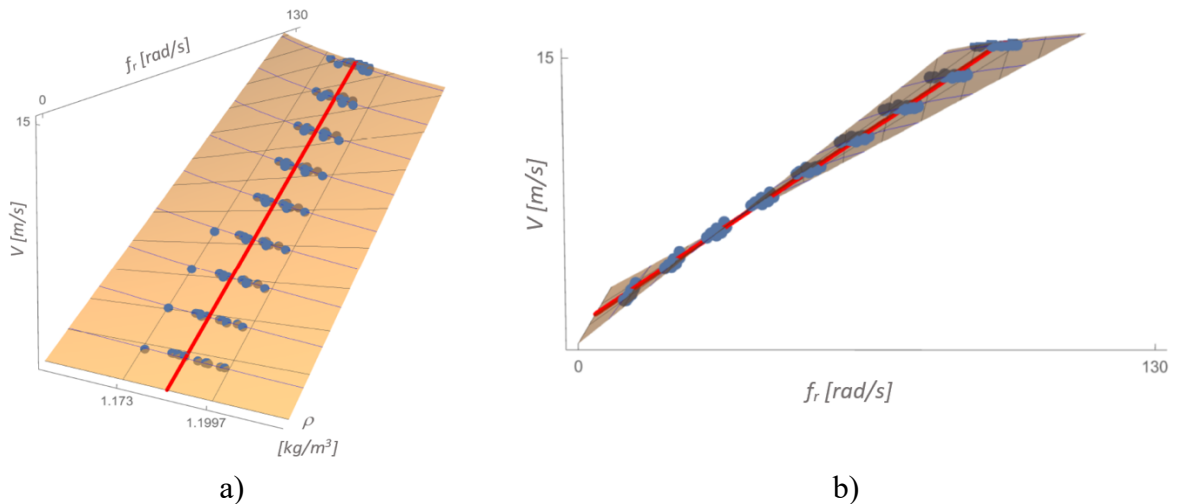
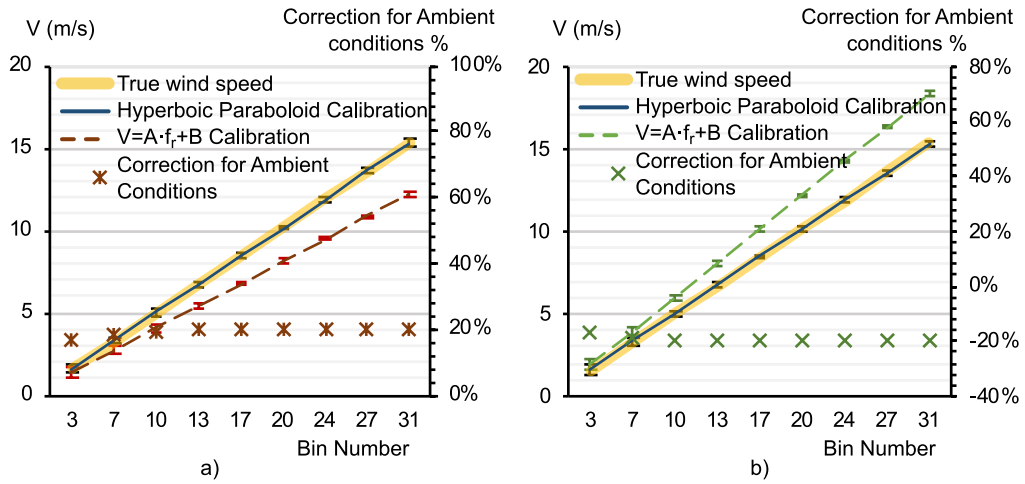


Fig. 11. Two views of the experimental observations fitting: the surface is the hyperbolic paraboloid that take into account ambient conditions, and the highlighted line is obtained from the IEC anemometry method [11].

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314 **Fig. 12.** Example of Direct Method and Dimensionless Abacus Method regressions models for an Auriol IAN cup anemometer.  
315 a)  $\rho_c = 1.1863 \text{ kg/m}^3, \rho = 1.225 \text{ kg/m}^3$ , b)  $\rho_c = 1.1863 \text{ kg/m}^3, \rho = 1.1476 \text{ kg/m}^3$ .

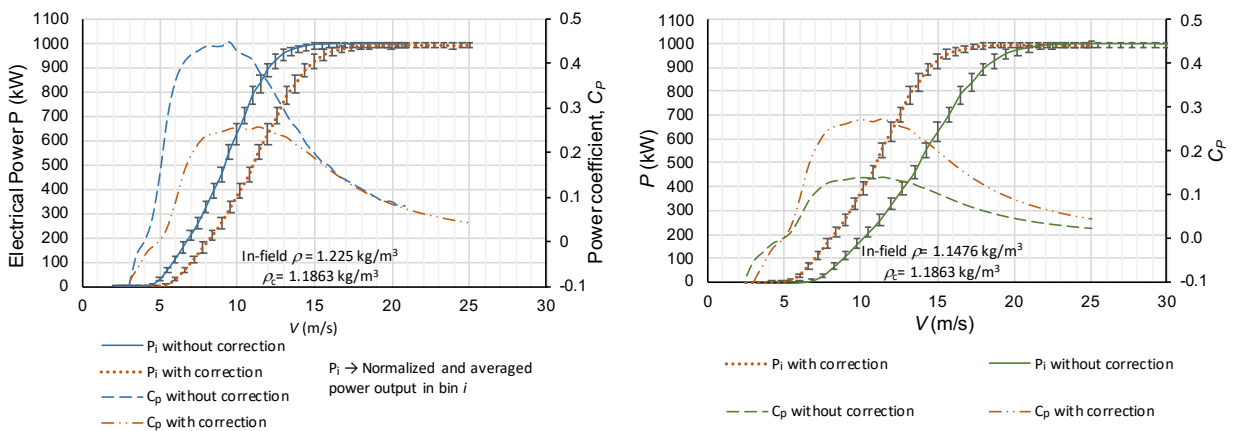
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317 **3.3. Annual Energy Production.**

318 Annual Energy Production AEP, which depends on the turbine power curve and the wind speed probability distribution during a  
319 year, is key to study turbines performance.

320 The power curve relates wind speed and output power. According to the bin method described in the IEC 61400-12-1 international  
321 standard [1], an average value of wind speed and power corresponds to a specific bin (wind speed  $V_i$ - electrical power  $P_i$ ) of the  
322 power curve. This method does not consider the influence of density variations between field and calibration laboratory on power  
323 curves.

324 A study case is presented to understand the role of the aforementioned density variations in power curves definition. In this case,  
325 the power data and wind speed measures provided in page 67 of the IEC 61400-12-1 standard [1] are used considering that wind  
326 speed was obtained by means of the analysed Auriol anemometer. Fig. 13 shows the resulting power curves. If the in-field air  
327 density  $\rho$  is greater than that at the calibration laboratory  $\rho_c$ , then the corrected power curve is beneath the power curve provided  
328 by the bin method (Fig. 13.a) and, on the contrary, when  $\rho_c > \rho$  the bin method curve is beneath the corrected power curve (Fig.  
329 13.b). Note that, even small density variations ( $\pm 0.04 \text{ kg/m}^3$  in Fig. 13) lead to significant differences on the power curves (up to  
330 20% of relative error, see Fig. 13). Actually, kinematic viscosity is the responsible of the cup anemometer rotation speed  
331 variations and therefore, it should be the parameter for correcting the power curve. Nevertheless, as it was previously discussed,  
332 for small density variations there is a direct relationship between density and the inverse of the kinematic viscosity so that density  
333 can be used instead of viscosity to correct the power curve.

334 According to the observed results in Fig. 13, power curves should be corrected in order to compare the turbine performance at  
335 different locations with different densities. This correction involves applying Eq. 19, which depends on the anemometer used;  
336 for the studied Auriol anemometer the value of  $c$  coefficient is shown in Fig. 7.



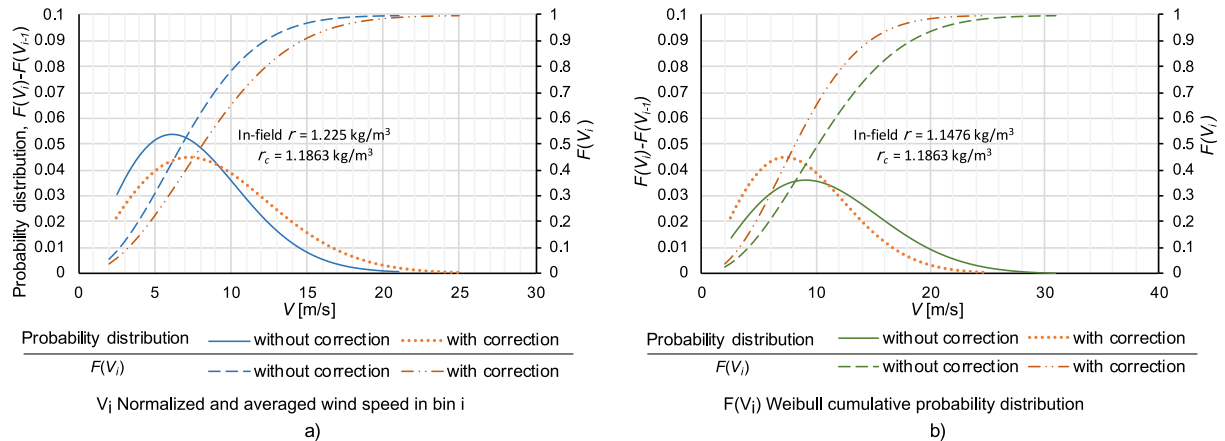
338  
339 **Fig. 13.** Study case: power curves and power coefficient  $C_p$  curves. a)  $\rho > \rho_c$  b)  $\rho_c > \rho$ . Confidence intervals with a coverage  
340 factor of one.

341 In addition to the power curve, it is necessary a wind speed probability distribution for AEP estimation. This probability  
342 distribution is also affected by the ambient conditions. The same method previously discussed can be applied to correct this  
343 distribution considering the density of a specific place and the used cup anemometer.

344 For the study case under consideration, Fig. 14 shows the resulting wind speed Weibull cumulative distribution  $F(V_i)$  and its  
345 probability distribution with and without density corrections at two locations with different air densities. For a place with  $\rho > \rho_c$ ,

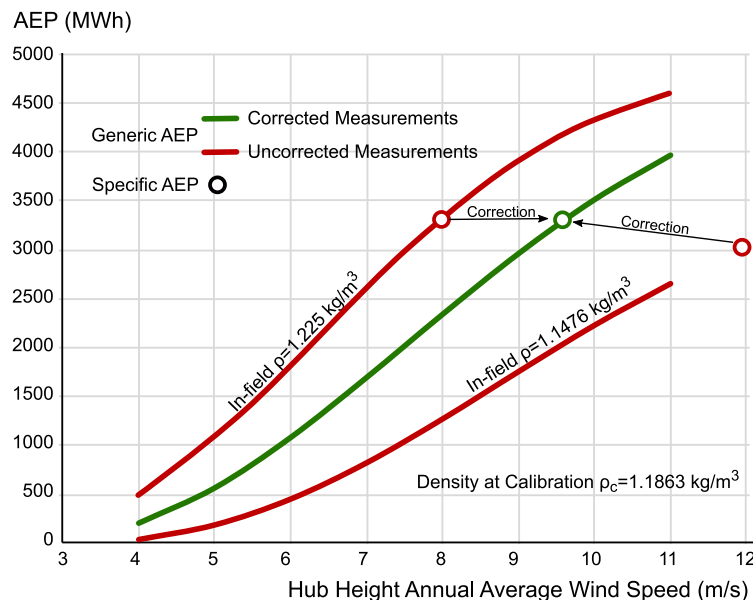
346 the viscous effects slow down the anemometer's rotational speed respect to the rotation speed observed during calibration. This  
 347 means that the frequency assigned to a wind speed without density correction should be assigned to a higher wind speed. On the  
 348 contrary, in locations with lower density, the viscous effects are reduced, which increases the anemometer's rotational speed and  
 349 therefore, the wind speed value provided by the anemometer is higher than the actual one. Fig. 14 also portrays the wind speed  
 350 frequencies with and without correction. The mode values of the wind speed frequency curves comply with the abovementioned  
 351 observations: higher mode value for the corrected curve at places with air density greater than the calibration one (Fig. 14.a) and  
 352 vice versa. Recall that in this study case there are small density variations, otherwise viscosity should be considered instead of  
 353 density.

354 It is worth to note that it is impossible to make rigorous comparisons between wind frequency curves of two different sites without  
 355 accomplishing corrections for ambient conditions. The average annual wind speed (or the Weibull scale factor) should be  
 356 corrected upwards at those places with air density above the calibration one (typically coastal locations), and otherwise it should  
 357 be corrected downwards (typically interior or mountainous locations).



358  
 359 **Fig. 14.** Study case: Weibull cumulative probability distribution and wind speed frequency distribution: a)  $\rho > \rho_c$  b)  $\rho_c > \rho$

360 Finally, according to the international standard IEC 61400-12-1 [1], specific AEP is calculated as the product of power curve and  
 361 the wind speed frequency distribution at a specific site, while generic AEP is obtained by multiplying measured power curve by  
 362 a set of reference distributions for wind speed frequency. Each reference distribution has a mean value of wind speed measured  
 363 at hub height. Fig. 15 shows specific and generic AEP measured using the power curves and the frequency distributions shown  
 364 in Figs. 13-14. Generic AEP is typically used for design optimization of wind turbines [29] and specific AEP is used for  
 365 determining the optimal location among several proposed places for the installation of a wind turbine [30].



366  
 367 **Fig. 15.** Example of specific and generic AEP using an Auriol IAN cup anemometer,  $\rho_c = 1.1863 \text{ kg/m}^3$ .

368 Regarding generic AEP, it is observed that the values at sites with  $\rho > \rho_c$  should be corrected downward; whereas when the  
 369 generic AEP is calculated in locations with  $\rho < \rho_c$ , the generic AEP should be corrected upwards. Therefore, when calculating  
 370 estimations of generic AEP, the effect of the variation in ambient conditions between the measuring location and the calibration  
 371 laboratory of the anemometer must be considered. The effect of air density can be corrected by means of Eq. (19). Nevertheless,  
 372 if the variation range of density is significant, the correction should be based on the kinematic viscosity, as it is stated in Eq. (22).

373 As explained above, the specific AEP for a particular location can be calculated from the mean power curve and the wind speed  
 374 specific frequency curve. If both curves are measured at the same time and using the same anemometer, the resulting value of  
 375 the specific AEP does not require a correction of the effect of air density during the calibration procedure (see Fig. 15). This is  
 376 because the correction affects both curves in an inversely proportional way, which leads the result of the multiplication to remain

377 unaffected. The problem arises from the fact that is necessary to measure the wind turbine power curve at its final location, or at  
378 a near one. Nevertheless, the research literature shows that the same anemometer is rarely used, due to the manufacturer power  
379 curve is used [30] or wind speeds are obtained from wind resource maps [31,32].

380 The proposed correction procedure allows to study the wind energy potential of a wind turbine before installing it in a specific  
381 location. The idea is to correct a known power curve of the wind turbine considering the variation of air density between a specific  
382 place where the turbine could be installed and the calibration laboratory.

383

#### 384 4. Conclusions

385 This work provides an explanation for those cases where a cup-anemometer has different rotation frequencies for the same wind  
386 speed. According to a set of experimental measurements conducted in a wind tunnel, we concluded that the main cause of the  
387 abovementioned variations are the kinematic viscosity variations or, equivalently, the density variations when a constant dynamic  
388 viscosity can be assumed. Moreover, we verified that the experimental observations fit adequately to a hyperbolic paraboloid  
389 surface, which relates wind speed, rotation frequency and density.

390 The proposed relation leads to more accurate wind speed measurements. Hence, it helps to improve the AEP estimations by  
391 reducing the error linked to the cup-anemometers calibration procedure. On the other hand, in view of the presented results, for  
392 those that prefer a conventional calibration (which is accomplished considering a constant density), we described a procedure to  
393 correct in-field wind speed measurements according the specific density conditions.

394 Respect to the observations that foster this work, even for a low-density variation of  $0.0267 \text{ kg/m}^3$ , we noted differences up to 20  
395 rd/s for a constant wind speed of 17 m/s as we showed in Figure 4. A linear regression between the slopes  $A$  of  $V/f_r$  calibration  
396 curves for different densities failed due to great dispersion of  $(\rho, A)$  data (Figure 5).

397 In order to overcome the above drawback, we discovered that representing the information in a  $\text{TSR}^{-1}$  vs  $\text{Re}$  diagram, it is possible  
398 to estimate the calibration slope coefficient with a relative uncertainty lower than 0.005, as a linear function of the density or the  
399 kinematic viscosity. The resulting relation among wind speed, rotation frequency and density is a type of ruled surface (a  
400 hyperbolic paraboloid surface). This surface, fitted using our experimental data, has a maximum absolute error of 0.4 m/s while  
401 the conventional calibration line has a maximum absolute error of 0.9 m/s. Experimental results show that, when field air density  
402 varies  $\pm 0.01 \text{ kg/m}^3$  from that at the calibration laboratory, the error in wind speed provided by the cup anemometer is about  $\pm 5\%$ .  
403 The conventional calibration can be improved by correcting the measure with  $\pm \Delta V$  that depends on the air density and the  
404 coefficient  $c$  (cup-anemometer air density sensibility) of the hyperbolic paraboloid surface (Eq. 19).

405 Regarding wind turbine power curves, if in-field measurements of wind speed with a cup-anemometer are not corrected, a 50%  
406 variation in the power coefficient  $C_p$  is observed when the same wind turbine is tested at several locations whose air density  
407 slightly differs from that at the calibration laboratory. Therefore, wind power curves for the same turbine differs significantly  
408 depending on the location where the observations are made. If the correction procedure proposed in the present study is applied,  
409 the estimated value of wind turbine power decreases as in-field air density increases respect to that of the calibration laboratory,  
410 and vice versa. Therefore, the proposed procedure provides the same power curves for a particular cup-anemometer regardless  
411 of the location where measurements are performed.

412 Additionally, using the power data and wind speed measures provided by the IEC 61400-12-1 standard for a 1 MW wind turbine,  
413 we computed the AEP with and without applying the developed equation for cup anemometers. The results of this computation  
414 showed for example, overestimations of 1000 MWh/year for an annual mean wind speed of 10 m/s measured at hub height in a  
415 location where air density is about  $0.04 \text{ kg/m}^3$  higher than that at the calibration laboratory, and underestimations about 1250  
416 MWh/year when density is  $0.04 \text{ kg/m}^3$  lower. The error of the estimation is even greater for lower wind speeds.

417 Just in case of determining the power curve and the wind speed specific frequency at the same location and with the same  
418 anemometer, specific AEP estimations don't require anemometer correction. Therefore, this case is feasible only when measuring  
419 an existing wind turbine.

420 Finally, the following bullets summarize the main findings of this work:

- 421 • The rotation frequency variations for constant wind speeds in cup-anemometers are due to kinematic viscosity  
422 variations.
- 423 • A hyperbolic paraboloid surface that relates wind speed, rotation frequency and kinematic viscosity provides a more  
424 accurate wind speed estimation than a linear regression between wind speed and rotation frequency at constant density.
- 425 • The accuracy of cup anemometers would improve significantly by considering the calibration error and, consequently,  
426 wind speed measurements would lead to more accurate AEP estimations."

427 The novelty and interest of the summarizing findings are that they quantitatively explain the observed rotation frequency  
428 variations in a cup anemometer respect to ambient conditions. The proposed approach would improve the reproducibility of the  
429 measurements provided by cup anemometers, which would increase the reliability of these devices when they are used to perform  
430 power curves, statistical studies regarding wind speed and estimations of AEP.

431 Since the proposed approach has been done using wind tunnel measurements and a specific anemometer, future works that would  
432 help to support the presented results could be to perform in field measurements with different anemometers, or do the cup-  
433 anemometer calibration in a wind tunnel that allows large variations in air density. For this last case, the wind tunnels must allow  
434 a regulation of air pressure, so that it can be slightly modified respect to the ambient pressure.

435

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